# THE DEVELOPMENT OF A MATHEMATICAL APPROXIMATION TECHNIQUE TO DETERMINE THE MASS MEDIAN AERODYNAMIC DIAMETER (MMAD) AND GEOMETRIC STANDARD DEVIATION (GSD) OF DRUG PARTICLES IN AN INHALATION AEROSOL SPRAY

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## ABSTRACT

aerodynamic diameter of an airborne particle is the key property in determining its respiratory deposition. may differ in their shape, size, and/or density and be aerodynamically indistinguishable because similar behavior in a moving airstream. A cascade impactor has been used to measure the range of aerodynamic diameters for drug particles in an aerosol spray product. The two terms associated aerodynamic particle sizing are Mass Median Aerodynamic Diameter (MMAD) and the Geometric Standard Deviation (GSD). are determined from the visual fit of a straight line to data manually plotted on log-probability paper. The plotting of this data is a time consuming activity and the fit of the line to the data can be very subjective.

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axis is an integral function not easily adaptable to the computer packages generally used in the laboratory. approximation has been developed that in effect "linearizes" the The mathematical relationship (a fifth degree probability axis. expression) when incorporated into a general computer allows for the direct computation of the two aerodynamic Since the subjectiveness of manually plotting the data terms. the line are removed, the results obtained Also, the results are obtained more potentially more accurate. quickly with significantly less effort. This paper explores the development of the mathematical relationship and provides of values obtained by both the manual and computer comparison methods.

#### INTRODUCTION

The human respiratory tract can be viewed as an aerodynamic classifying or particle sizing system for airborne particles. 1,2 particles may differ in shape, size, and/or density and yet aerodynamically indistinguishable because they have similar behavior in an airstream. Aerodynamic diameter is property for respiratory deposition of airborne particles. many cases, it is not necessary to know the true size, shape, or density of a particle if the aerodynamic diameter determined.3 The cascade impactor has been proposed aerodynamic sizing of drug particles in an inhalation aerosol spray.4

A cascade impactor has been used to aerodynamically size drug in an inhalation aerosol product. The device consists a series of separation plates (stages) with each representing The drug particles an increasingly smaller aerodynamic diameter. introduced into the system via a moving airstream. particles cascade through the increasingly smaller stages until they are impacted as a function of their aerodynamic diameters.



The amount of drug impacted on each stage is quantified relative percent of the total drug impacted on all of the stages. A cumulative percent value is then determined for each stage by adding the relative percent values for all stages aerodynamically than stage of interest. The two terms associated with aerodynamic particle sizing, Mass Median Aerodynamic Diameter (MMAD) and the GSD (Geometric Standard Deviation), are calculated manually plotting the cumulative percent data versus the aerodynamic diameter of each stage on log-probability paper and fitting a straight line. The plotting of this data is a time consuming activity and the fit of the line to the data is very Although this technique is an excellent candidate for computerization, it cannot be done directly since probability axis is an integral function not easily adaptable to the computer packages generally used in the laboratory.

A fifth degree polynomial expression has been found that can used to "linearize" the probability axis. This relationship when incorporated into a general computer program allows for the direct computation of the two aerodynamic terms. Results can now be obtained more quickly with much less effort. This paper will the development of the mathematical relationship and compare the values obtained to those obtained the manual plotting technique.

### BACKGROUND

a flowing stream of air, small spherical particles will have a lower inertia than larger spherical particles of the same If an obstacle is placed directly in the path of this flowing stream of air, the direction of the airstreams will be changed. Particles of high inertia will not be able to adapt to this change in direction and will impact upon the The Stokes equation<sup>5</sup> is used to calculate obstacle.



aerodynamic diameter (also called Effective Cutoff Diameter associated with the obstacle. This obstacle serves as a of airborne particles into two sizes; separator particles larger than the aerodynamic diameter of impacted are separator whereas particles that are not impacted are smaller aerodynamic diameter parameter associated with separator. It is not the intent of this paper to provide a detailed discussion of impactor theory. A more detailed discussion of this topic can be found elsewhere in the references cited.

A cascade impactor consists of a series of up to eight stages with a collection disc positioned directly below each stage.6 The collection disc serves as the obstacle to change the direction of the airstreams. Each stage represents increasingly smaller aerodynamic diameter. Drug particles cascade down through the unit and are sized according The amount of drug aerodynamic diameter. found collection disc is determined and the mass of drug distributed over a range of aerodynamic diameters is then calculated. this data, the MMAD and GSD are obtained.

# MMAD AND GSD USING LOG PROBABILITY PAPER

drug impacted on each collection disc The amount of and a cumulative percent distribution of the drug is The plotted manually determined. data obtained islog-probability graph paper. The aerodynamic diameter or ECD (in microns) for each stage of the cascade impactor is plotted on the logarithmic scale axis versus the cumulative percent on the probability scale axis. The cumulative percent term represents amount of drug (as a percent of the total amount of drug found on collection discs) for all of the stages with aerodynamic diameters less than the stage of interest. typical data obtained for an eight stage cascade impactor.



TABLE 1 Typical Cascade Impactor Data

STAGE #	Cumulative %	ECD(microns)
Preseparator	98.0	9.00
1	95.0	6.20
2	92.0	4.37
3	75.0	2.99
4	50.0	2.02
5	20.0	1.09
6	12.0	0.70
7	5.0	0.52

Figure 1 is a plot of this data on log-probability paper. calculate the aerodynamic terms a straight line is visually fit to the data points. The MMAD is obtained by determining the aerodynamic diameter (ECD) that corresponds to 50%. diameter (ECD) corresponding to 15.9% (D<sup>15.9%</sup>) aerodynamic 84.1% ( $D^{84.1}$ %) are determined. For a normal distribution, one standard deviation is equal to the difference between either the aerodynamic diameter at 50% and the aerodynamic diameter at 84.1% (or the difference between 15.9% and 50.0%). For a log-normal which is normal with aerodynamic distribution respect to diameter8, the geometric standard deviation becomes:

$$GSD = \sqrt{\frac{D^{84 \cdot 1}}{D^{15 \cdot 9}}}$$

GSD is a unitless term which can never be less than 1.0 and is only attainable when all particles have exactly the same aerodynamic diameter. The value obtained from the graph for MMAD 1.8 microns. The values for  $D^{84.1}$  and  $D^{15.9}$  are 0.85 and 3.8 microns, respectively. The GSD calculated is 2.11.



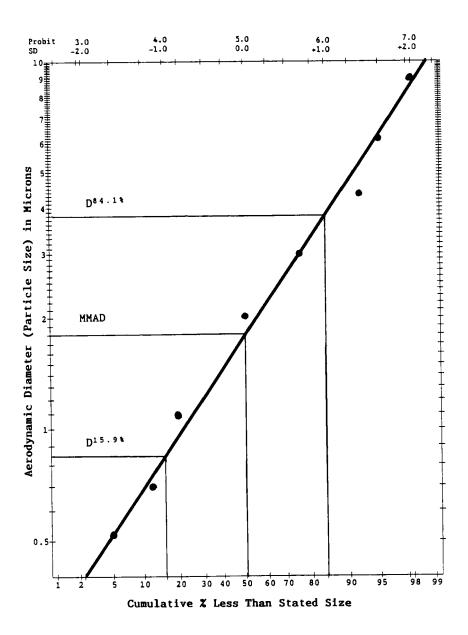


FIGURE 1 Manual Plot of Data On Log-Probability Paper



TABLE 2 Same Data in Tabular Probit and Log10 ECD Terms

STAGE #	Tabular Probit	Log ECD
Preseparator	7.06	0.954
1	6.65	0.792
2	6.41	0.640
3	5.67	0.476
4	5.00	0.305
5	4.16	0.037
6	3.83	-0.155
7	3.35	-0.284

# MMAD AND GSD USING TABLE OF PROBIT VALUES

50% point on the cumulative percent (probability) axis represents the mid-point of a normal distribution. 84.1% on the same axis is one standard deviation (+1.0) from this The point at 15.9% is one standard deviation (-1.0)in the opposite direction from the mid-point. A probit term can be used to represent the linear displacement from the mid-point along this axis. The value assigned to a probit is the standard deviation term plus 5. In probit terms, the point at 15.9% is the point at 50% is 5.0, and the point at 84.1% is 6.0. using a Table of Probits9, it is possible to assign any cumulative percent value a tabular probit value.

computer graphic packages require that both axes to be linear if the fit of a straight line to the data is desired. The use of probit values in effect "linearizes" the cumulative percent axis. The aerodynamic diameter axis can be linearized by the the logarithm of the ECD term. simply taking contains the same data as Table 1 with the cumulative values assigned the tabular probit equivalents and the logarithm of the ECD values substituted for the ECD values.



Figure 2 is the graph created from the tabular probit log10 ECD data contained in Table 2.

fit of a straight line to the data is then made by the computer.

The resulting equation for the line fit is: y = mx + b

where:  $y = \log_{10}$  of aerodynamic diameter for the stage

x = probit value corresponding to the cumulative %

b = y-intercept

m = slope of line

From the slope and intercept of the line the D15.9%, MMAD and can be calculated. The MMAD occurs where the cumulative percent is 50% or when x equals 5. When x = 5, y = 5m+b and represents the  $\log_{10}$  of the aerodynamic diameter at 50%, the MMAD becomes  $10^{5\,\text{m+b}}$ . At 15.9% cumulative percent, x = 4 thus y =The aerodynamic diameter at 15.9% is  $10^{4m+b}$ . At 84.1% cumulative percent, x = 6 so y = 6m+b. The aerodynamic diameter at  $D^{84.1\%}$  is  $10^{6m+b}$ . For this example,  $D^{15.9\%} = 0.87$  microns, MMAD = 1.82 microns, and  $D^{84.1\%} = 3.82$  microns. The GSD calculated as 2.10.

This approach is a major improvement over the manual plotting technique but it still requires looking up probit values in a table.



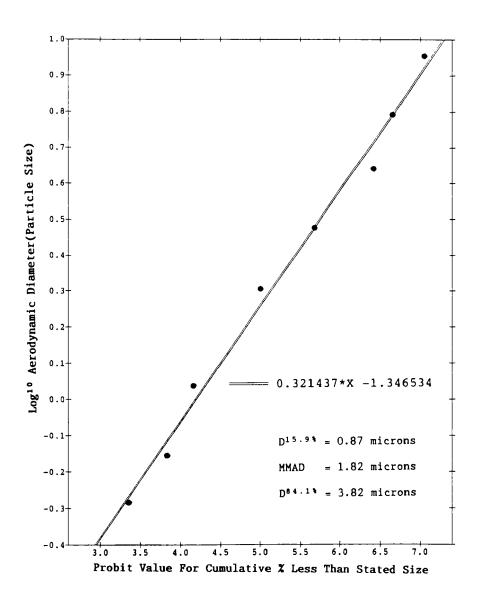


FIGURE 2 Computer Plot of Data Using Tabular Probits



# MMAD AND GSD USING CALCULATED PROBIT VALUES

probability (cumulative percent) axis is a normal The distribution represented by the following integral expression:

$$P = \int_{2\pi}^{1} \int_{-\infty}^{y-5} e dt$$

where (represented by the probit):

Attempted use of this integral function in a computer program was viewed as undesirable. A plot of tabular probit values versus the corresponding cumulative percent was made in an attempt if a more quantifiable relationship could determine Through the empirical fit of a function to the data it was found that a fifth degree polynomial expression could be used to transform cumulative percent to probit values which can plotted on a linear axis. The expression obtained through this technique is:

CPV = 
$$4.347524e^{-0.9}x^5 - 1.086881e^{-0.6}x^4 + 1.046449e^{-0.4}x^3$$
  
- $4.827924e^{-0.3}x^2 + 0.133068x + 2.770212$ 

CPV = the calculated probit value where: x = the cumulative percent value

Figure 3 shows a fit of the function to the tabular probit values that correspond to the cumulative percent. Comparing the tabular probit values with the calculated probit values over range of 2 % to 98 % (see Table 3) shows very good agreement.



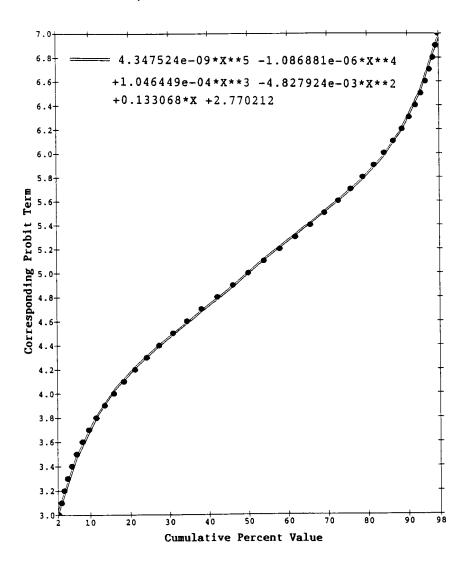


FIGURE 3 Plot of Cumulative Percent Versus Tabular Probits

in the previous example, a graph can be constructed by plotting the logarithm of the aerodynamic diameter(ECD) for each stage versus the probit term. However, instead of assigning a probit term to each cumulative percent value from a probit table, can now be directly calculated using the mathematical Table 4 contains the same data as Table 2 except relationship.



TABLE 3 Comparison of Calculated and Tabular Probits

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Percent	Tabular	Calculated	Difference
98.00	7.06	6.98	0.08
97.72	7.00	6.95	0.05
96.41	6.80	6.81	-0.01
94.52	6.60	6.63	-0.03
91.90	6.40	6.42	-0.02
88.50	6.20	6.20	0.00
84.10	6.00	5.98	0.02
78.80	5.80	5.78	0.02
72.60	5.60	5.60	0.00
65.50	5.40	5.42	-0.02
57.90	5.20	5.22	-0.02
50.00	5.00	5.00	0.00
42.10	4.80	4.78	0.02
34.50	4.60	4.58	0.02
27.40	4.40	4.40	0.00
21.20	4.20	4.22	-0.02
15.90	4.00	4.00	0.00
11.50	3.80	3.80	0.00
8.10	3.60	3.58	0.02
5.48	3.40	3.37	0.03
3.59	3.20	3.19	0.01
2.28	3.00	3.05	-0.05
2.00	2.95	3.02	-0.07

that the probit values have been calculated using the polynomial expression.

The fit of a straight line to the data by the computer is again used to calculate the slope and intercept. Figure 4 is a plot of the data from Table 4 using calculated probit values. For this example,  $D^{15.9\%} = 0.87$  microns, MMAD = 1.83 microns, and  $D^{84.1\%} = 3.84$  microns. The GSD is calculated as 2.10.



TABLE 4 Same Data in Calculated Probit and Log<sub>10</sub> ECD Terms

STAGE #	Calculated Probit	Log ECD
Preseparator	6.98	0.954
1	6.67	0.792
2	6.42	0.640
3	5.67	0.476
4	5.00	0.305
5	4.18	0.037
6	3.83	-0.155
7	3.32	-0.284

# EVALUATION OF THE FIT OF THE FUNCTION

has already been shown that the calculated probit values excellent agreement with the table probit values. However, there is some divergence at the tails below 2% and above This divergence occurs because probit values calculated 98%. the fifth degree polynomial expression never exceed 2.77 and 7.33 (100%). Upper and lower limits are not attainable using a probit table because the distribution is infinite. divergence is a minor issue since it would not be wise to place a significant weighting on these extreme values when generating the graph used to calculate MMAD and GSD. In fitting the line to the data the heaviest weighting should be given to cumulative percent values of between 20% and 80%.10

# SUMMARY

same data set was examined by the three calculation approaches discussed. The aerodynamic terms calculated



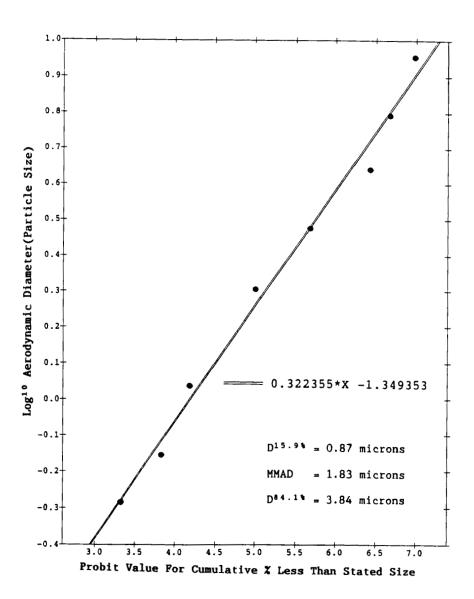


FIGURE 4 Computer Plot of Data Using Calculated Probits



TABLE 5 Comparison of Aerodynamic Terms Obtained

	Manual Plot	Tabular Probit	Calculated Probit
MMAD	1.80	1.82	1.83
GSD	2.11	2.10	2.10

summarized in Table 5. The results obtained by all approaches are in very good agreement.

The major benefit derived in using calculated probit values is that the mathematical expression used to calculate them can be incorporated a general computer program. The fit of a line to the data by the computer allows for the direct computation of the two aerodynamic terms from the slope and intercept values. tedious and time consuming activity of manually plotting the data is eliminated. The manpower effort required to obtain the results is significantly reduced.

In the example studied, the fit of a line to the data was not very challenging. Most of the time this is not the case and the visual of a straight line to the data becomes very The use of a computer to fit a line to the data eliminates this subjectiveness. Since both aerodynamic terms are dependent upon the fit of the line to the data, potentially more accurate results are now attainable using the approximation relationship.

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